Flight Test Results of the Head-Up Synthetic Vision Display

For the Quarterly Review of the NASA/FAA Joint University Program for Air Transportation Research Friday October 18th, 2002

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Ohio University, Athens
Project Sponsor: Joint University Program



Introduction

- General Aviation instrumentation has undergone little change in the past 50 years.
- Major advancements have been made in the areas of inertial navigation and high accuracy GPS.
- VMC into IMC flight continues to be one of the two major areas producing the largest number of GA fatalities.
- On average one spatial disorientation accident occurred every eleven days from 1987 to 1996.



Overview

- Motivation Behind Head-Up Synthetic Vision Display (HUSVD)
- Modern Synthetic Vision Systems vs. Prototype Display System
- Prototype HUSVD System Overview
- Flight Tests
- HUSVD New Architecture
- Attitude and Heading Reference System (AHRS)
- Multi-View HUSVD



No Longer Known as the eHUD

- Enhanced Vision Often Refers to a System that Incorporates Some Type of Infrared Sensor or Thermal Imaging Sensor to Augment a Virtual Display
- Head-Up Display or HUD Often Refers to a Collimated Symbolic Display
- The Head-UP Synthetic Vision Display (HUSVD) System is Neither Enhanced or Collimated

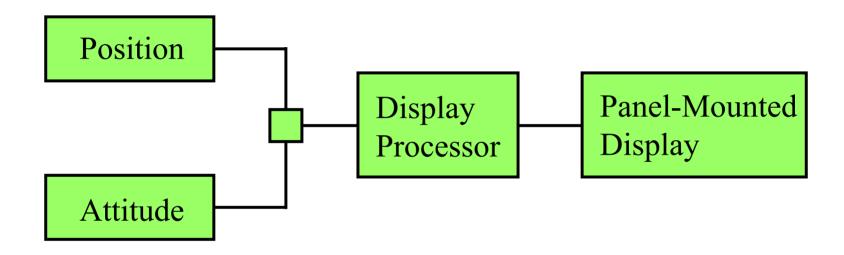


Motivation Behind HUSVD

- Provide Visual Cues in IMC
- Increase Situational Awareness in IMC
- Reduce Pilot Training and Currency Requirements for Flight in IMC
- Reduce Instrument Fixation During the Approach
- Cost-Effective Head-Up Synthetic Vision Display



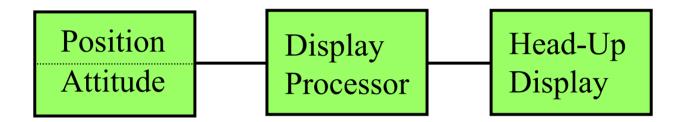
Typical Synthetic Vision System Architecture



- Aircraft Attitude Sensor
- Aircraft Position Sensor
- Display Processor
- Panel-Mounted Display



Prototype HUSVD System Architecture



- Attitude/Position Sensor
- Head-Up Display

Data Processor

The prototype HUSVD was flight tested on th 18th of July 2002



Pseudo-Attitude Determination

(Velocity Vector Based Attitude Determination)

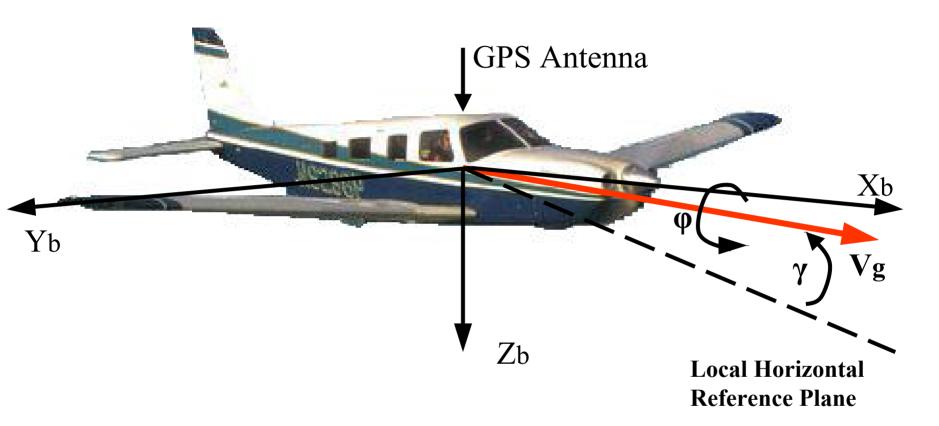
Developed at the Massachusetts Institute of Technology by:

- Dr. Richard P. Kornfeld
- Dr. R. John Hansman
- Dr. John J. Deyst

The information on the following slides, regarding Velocity Based Attitude, was taken from "*The Impact of GPS Velocity Based Flight Control on Flight Instrumentation Architecture*" Report No. ICAT-99-5, June 1999.



Pseudo-Attitude



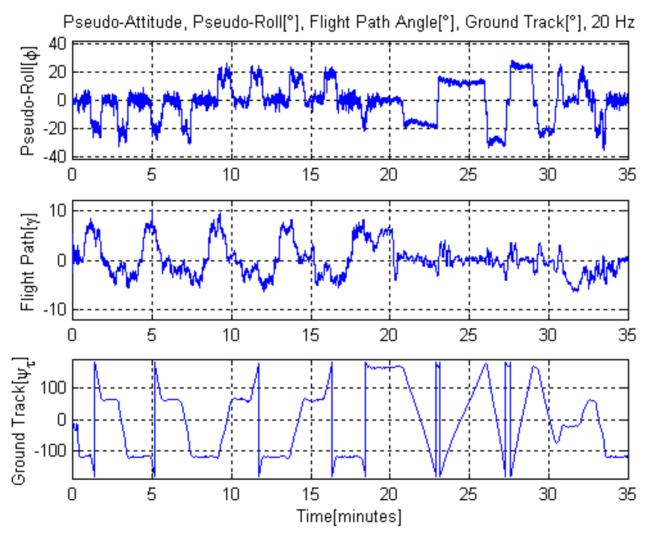
Flight Path Angle : γ

Pseudo-Roll Angle : φ

FB: Body-fixed orthogonal axes set which has its origin at the aircraft center of gravity.

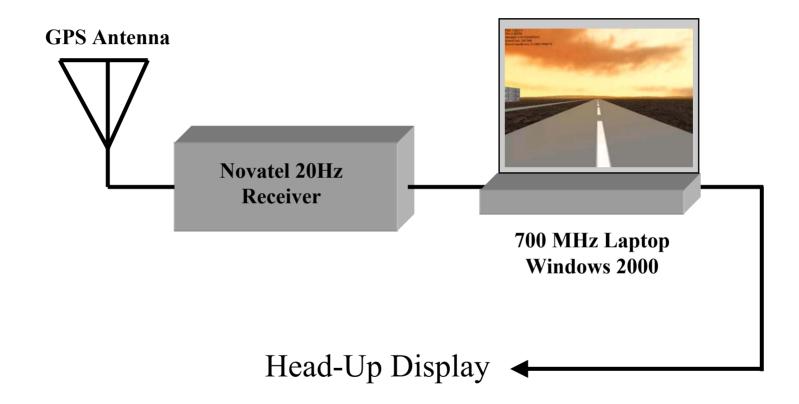


Pseudo-Attitude Over Time

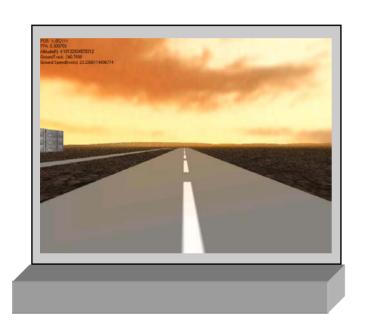




Current eHUD Configuration



Data Processor and Display Processor

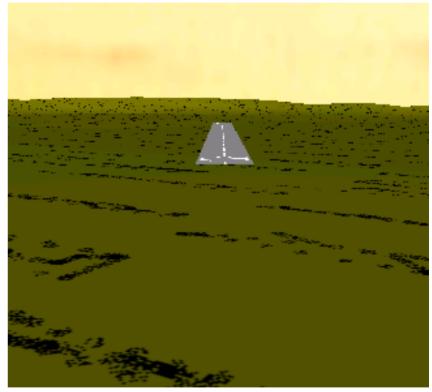


- 700 MHz Laptop Running Windows 2000
- Attitude Determination Algorithm
 Performed in C++ DLL
- Display Written in Visual Basic
- Graphics Produced Using Revolution 3D Graphics Engine
- Three-Dimensional representation of the outside world



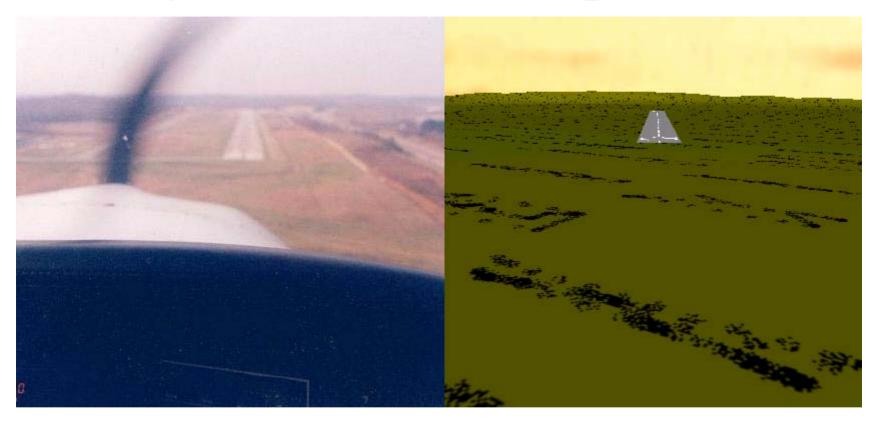
Image Layers







Synthetic Vision Comparison



Two separate test flights on UNI Runway 25. There is a slight altitude difference between the two approaches. Synthetic perspective is very compelling.



Flight Test Aircraft





Combiner Material: Lexan 9034





Projector Mounting





Equipment Installation







HUSVD: Climb-Out



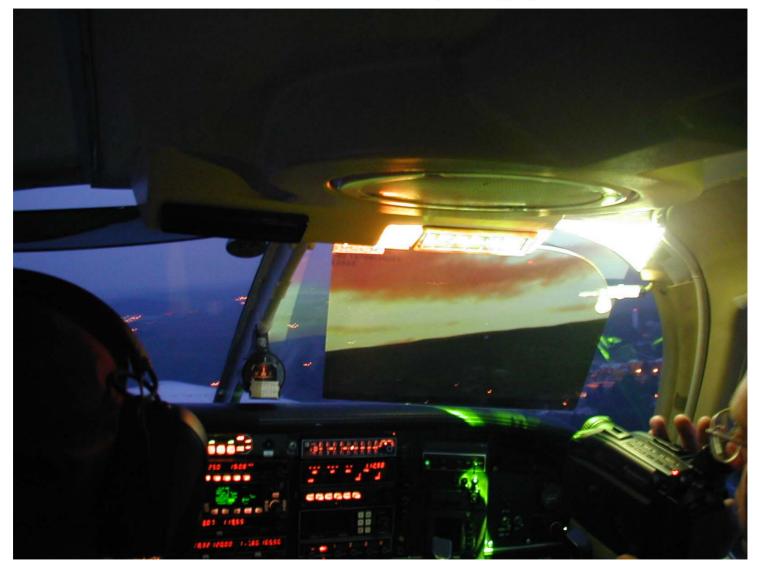


HUSVD: Down-Wind Leg

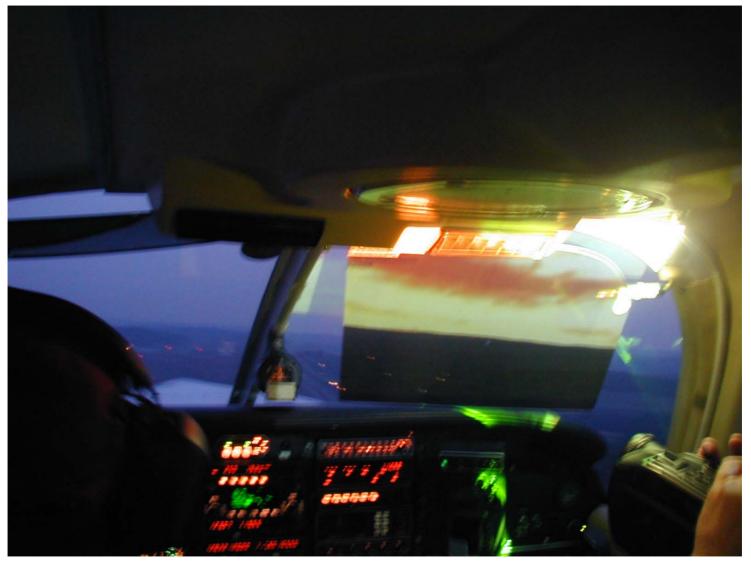




HUSVD: Shooting Approach







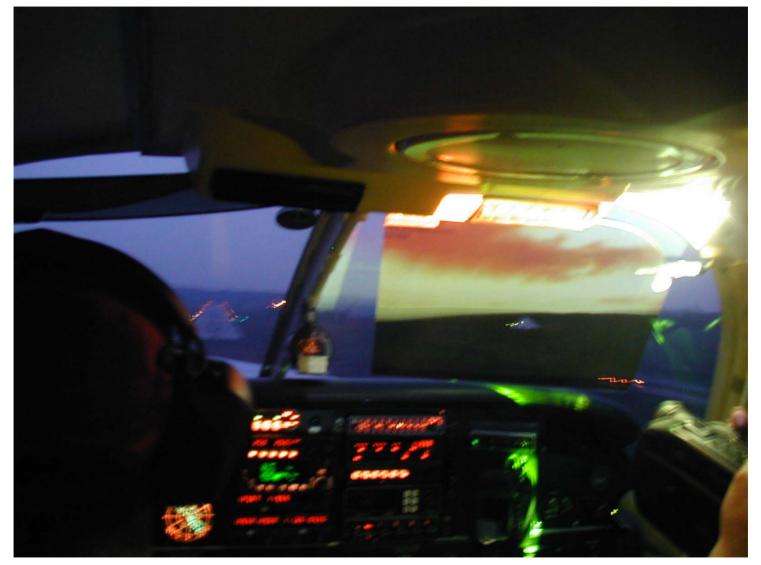






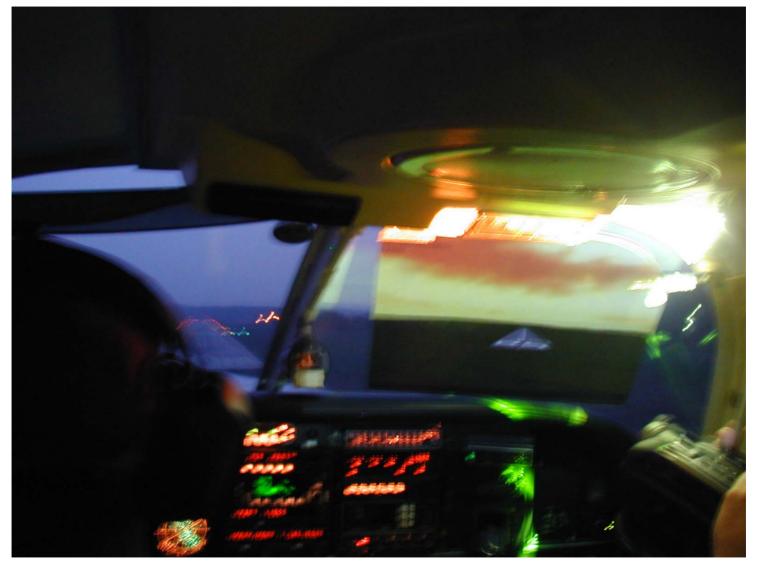








HUSVD: Before Touch-Down





HUSVD: Touch-Down



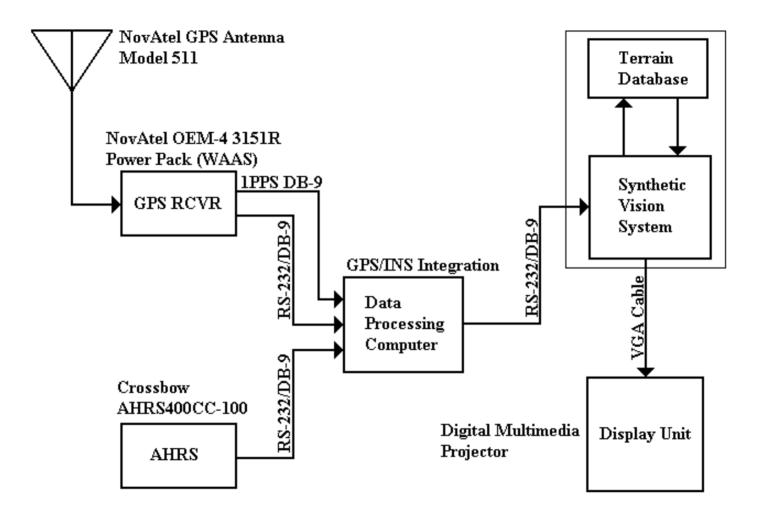


Lessons Learned

- Discrepancy in Size or Object Placement is an Immediate Distraction
- Synthetic Vision Head-Up and Head-Down Display Perspectives are Very Different
- Dynamic "Tuning" of HUSVD is Needed
- Object Size Needs to be Increased
- Lexan 9034 Promising Combiner Material
- LCD Projector Provides Navigation Quality Image in Various Lighting Conditions



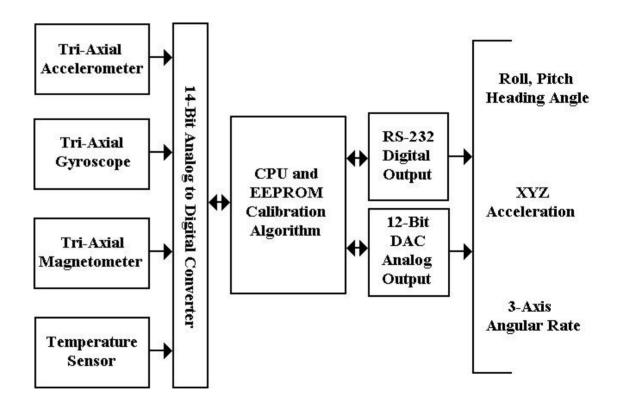
New System Architecture





Crossbow AHRS400CC-100



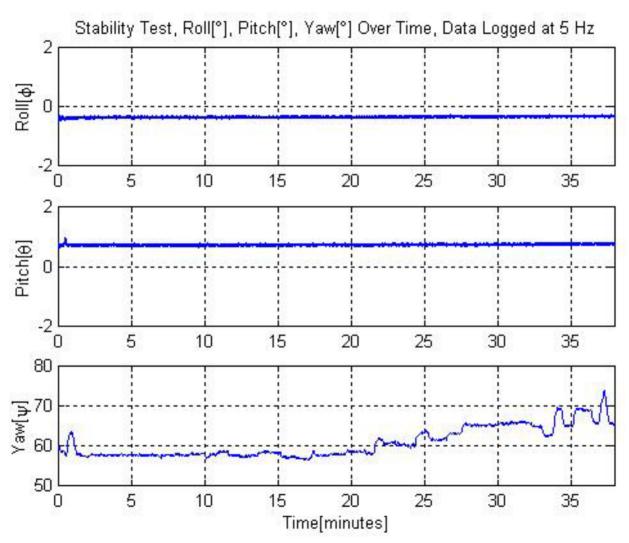


http://www.xbow.com





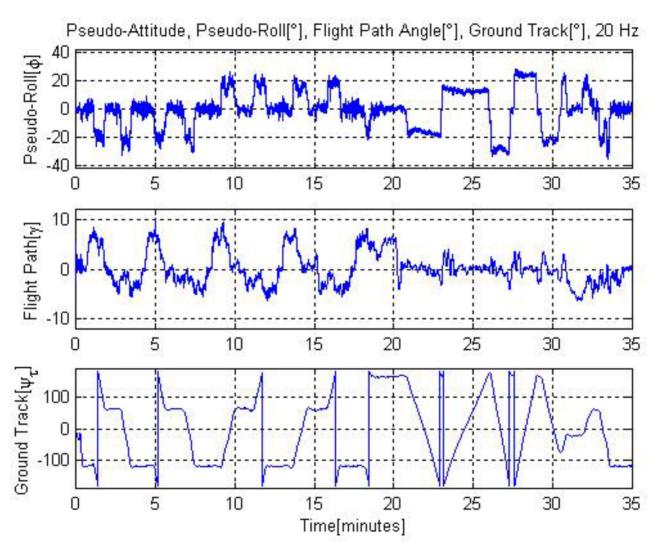
AHRS Static Evaluation







Velocity Vector Pseudo-Attitude



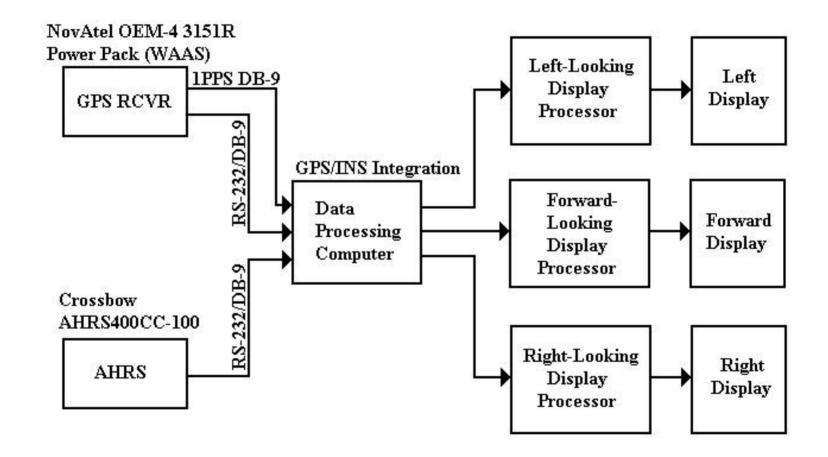


AHRS/GPS Integration

- AHRS Provides Robust Roll and Pitch
- AHRS Roll and Pitch will Drift Over Time Creating a Bias
- Complementary Kalman Filter will be used to Integrate the Single-Antenna GPS Attitude with the AHRS Measurements
- GPS Ground Track will be Used to Approximate the Yaw



Multi-View Head-Up Synthetic Vision Display System



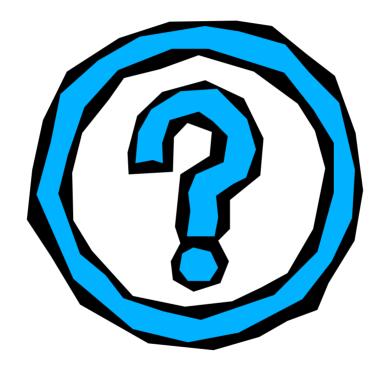


Current Work

- Expand System to Include Both Forward-Looking and Left-Looking Views
- Flight Test in which Multi-View HUSVD is used to Navigate Aircraft from Climb-out to Approach
- Upgrade Synthetic Vision Software



Questions





References

- Kornfeld, R.P., Hansman, R.J., Deyst, J.J., *The Impact of GPS Velocity Based Flight Control on Flight Instrumentation Architecture*. MIT International Center for Air Transportation, Cambridge, MA. Report No. ICAT-99-5, June 1999.
- Jennings, C., Alter, K.W., Barrows, A.K., Per Enge, J., D. Powell, 3-D Perspective Displays for Guidance and Traffic Awareness. Presented Sep 1999 at the ION GPS Conference, Nashville, TN.
- 1999 Nall Report, AOPA Air Safety Foundation, http://www.aopa.org
- Crossbow Technology, Inc. *AHRS400 Series User's Manual*, 41 E. Daggett Dr., San Jose, CA 95134, http://www.xbow.com



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